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TYPE: Magni M24 C

(1) MANUFACTURER: Magni Gyro Srl

Besnate Varese 21010 Italy

(2) UK IMPORTER Magni Gyro UK

AND DISTRIBUTOR: Chiltern Air Park

Ipsden Wallingford OXON OX10 6AS

UK

(3) CERTIFICATION: BCAR Section T Issue 3 (CAP 643)

(See AAN 29276)

BCAR Section T Issue 5 (CAP 643)

(M24C Plus, i.e. M24C equipped with 915iS

engine) (Addendum 1, Mod. No. 125)

BCAR Section T Issue 5 (CAP 643), CS-27 Amendment 7, and Air Navigation Order

Schedule 5 (M24C Plus Day/Night VFR variant,

i.e. M24C Plus with additional equipment)

(Addendum 2, Mod. No. 151)

(4) DEFINITION OF M24C Type Design Build Standard

BASIC STANDARD: 002-00-24C – Issue F or later approved revision

253-00-24C – Issue A or later approved revision (915iS engine only) (Addendum 1, Mod. No. 125 and Addendum 2, Mod. No. 151 Day/Night VFR variant)

(5) COMPLIANCE WITH GYROPLANE DEFINITION:

a) MTOW 500kg

535 kg (915iS engine only) (Addendum 1, Mod. No. 125)

b) No. Seats 2

c) Permitted range of pilot weights

Left Seat 60-120 kg Right Seat 120 kg max

Permitted total Occupant Weight: 203 kg (subject to fuel loading)

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		220 kg (subject to fuel loading) (915iS engine only) (Addendum 1 Mod. No. 125)
		218 kg (subject to fuel loading) (Day/Night VFR variant) (Addendum 2 Mod. No. 151)
d)	Typical Empty Weight (ZFW)	297 kg
		315 kg (915iS engine only) (Addendum 1 Mod. No. 125)
		317 kg (Day/Night VFR variant) (Addendum 2 Mod. No. 151)
e)	ZFW +180 kg crew + 1 hr fuel	496 kg
	(27 litres, 19 kg)	514kg (915iS engine only) (Addendum 1 Mod. No. 125)
		516 kg (Day/Night VFR variant) (Addendum 2 Mod. No. 151)
f)	ZFW + 90 kg crew + full fuel	449 kg
	(82ltrs, 62 kg)	464 kg (915iS engine only) (Addendum 1 Mod. No. 125)
		466 kg (Day/Night VFR variant) (Addendum 2 Mod. No. 151)
g)	Max ZFW at initial permit issue	301 kg 320 kg (915iS engine only) (Addendum 1 Mod. No. 125) 320 kg (Day/Night VFR variant) (Addendum 2 Mod. No 151)

(6) POWER PLANTS

Designation	M24C	M24C Plus (Mod. No. 125)
Engine Type	Rotax 914 UL	Rotax 915iS
Reduction Gear	2.43:1	2.54:1
Exhaust System	Rotax stainless steel with after muffler	Rotax stainless steel with after muffler
Intake System	Single intake filter and balance box	Single intake filter, fuel injected
Propeller Type	MAGNI GYRO ECO GL170/3 light ground adjustable composite propeller	DUC Hélices Inconel Flash- 2 4-blade ground adjustable propeller

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Propeller Diameter x Pitch	1.70 m x12°45' at tip of blade, with inclinometer against intrados of aerofoil 1.72 m x 17° at tip of b with inclinometer again intrados of aerofoil	
Noise Type Cert No.	None Required	None Required
AAN Approving Configuration	AAN 29276	AAN 29276 Addendum 1
ROTOR SYSTEM		

(7)

Rotor system description	Single two blade fixed pitch teetering rotor of 28 ft diameter.
AAN approving rotor system	AAN 29276

(8)**MANDATORY LIMITATIONS:**

(A) MTOW 500kg

> 535 kg (915iS engine only) (Addendum 1 Mod. No. 125)

(B) CG Limits

Horizontal CG Fwd: 1409 mm aft of datum

> Aft: 1589 mm aft of datum

Aft: 1,608 mm aft of the datum

(915iS engine only) (Addendum 1 Mod. No. 125)

Vertical CG Upper: 688 mm above datum

> Upper: 750 mm above the datum (915iS engine only) (Addendum 1, Mod. No. 125)

Lower: 518 mm above datum

(C) CG Datum:

Horizontal cg: Nose wheel fork stem axis Vertical cg: Keel beam upper surface

(D) Cockpit Loadings

Left Seat Min 60 kg

Max 120 kg

Right Seat Min 0 kg

> Max 120 kg

Total Max 203 kg

> Max 220 kg (915iS engine only) (Addendum 1, Mod. No. 125)

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Max 218 kg (Day/Night VFR variant) (Addendum 2, Mod. No. 151)

(E) Never Exceed Speed, V_{NE} 100 mph(F) Minimum Speed 30 mph

(G) Prohibited Manoeuvres: Aerobatic manoeuvres are prohibited.

Intentional spinning is prohibited.

Manoeuvres involving a deliberate reduction in normal 'g' shall be avoided. Flight into known icing conditions is prohibited (not placarded). Maximum crosswind 25 kts.

(H) Other Limitations Day VFR, or

Day/Night VFR (Addendum 2, Mod. No. 151)

This Gyroplane is hereby granted a permission to fly for the purposes of aerial work which consists of the giving of instruction in flying or the conducting of flying tests subject to the installation of any required instructional modifications and the Gyroplane being owned or operated under arrangements entered into by a flying club of which the person giving the instruction or conducting the test and the person receiving the instruction or undergoing the test are

both members.

(I) Fuel Contents: 82 litres (78.5 litres useable)

(J) Power Plant

Engine	Rotax 914 UL	Rotax 915iS
Max RPM	5,800	5,800
Max Continuous RPM	5,500	5,500
Max CT	120 °C	120 °C
Max EGT	950 °C	950 °C
Fuel Spec	MOGAS or AVGAS	MOGAS or AVGAS
Engine Oil Spec	Shell Advance Ultra – Fully Synthetic	AeroShell SPORT PLUS 4 Rotax® RON 424
Gearbox Oil Spec	Integral with Engine	Integral with Engine
Fuel/Oil Mixture	N/A	N/A
Coolant Temperature	N/A	N/A

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Oil Pressure	Max: 7 bar	Max: 7 bar
	Min: 1.5 bar	Min: 1.5 bar
	Normal operating 2 - 5 bar	Normal operating 2 - 5 bar
Oil Temperature	Max: 130 °C	Max: 130 °C
	Min: 50 °C	Min: 50 °C
Fuel Pressure	N/A	N/A
Manifold pressure (take-of	f) 39.9 inHg	51 inHg (managed by the
Manifold pressure (max	35.4 inHg	ECU)
continuous)		

(9) INSTRUMENTS REQUIRED:

ASI: Altimeter: Rotor Engine Compass: VSI: CHT: MAP: Fitted Fitted RPM: RPM: Fitted Fitted Optional Fitted Fitted Fitted

For Night VFR operations the aircraft has to be additional equipped with:

- LED landing light
- Heated pitot
- Cabin light
- Internal LED lights
- Position/navigation /strobe lights
- Tail position/strobe light protection
- Anticollision light
- Backlit Altimeter and Air Speed Indicator
- Backlit rotor rpm tachometer and compass
- Longer instrument panel eyelid
- Dedicated warning LED lights
- Dedicated breakers and switches panel
- Dedicated engine instrument panel with Dynon Avionics SV-HDX800 used as EMS
- Internal cabin static port system
- Portable torch for each crew member
- Strobe flashing protection
- Dynon screen protection
- Upper LED stripe protection

(10) CONTROL DEFLECTIONS:

Rotor Head – Roll: Rotor Head – Pitch: Rudder Deflection: 16° Total 19° Total Left: 40° ±5° Total Right: 45° ±5° Total

(11) PILOTS NOTES, MAINTENANCE MANUALS REFERENCES:

- 11.1 Manuals approved for use with this aircraft;
 - (a) Flight Manual 025-00-24C Issue H or later approved revision.

For the 915iS variant M24C Plus Flight Manual 227-00-24C Issue B or later approved revision.

For the M24C Plus Day/Night VFR variant (Mod. No. 151) Flight Manual 363-00-24C Issue A.02 or later approved revision.

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(b) Maintenance Manual 026-00-24C Issue D or later approved revision.

For the M24C equipped with the Rotax 915iS (M24C Plus) and Day/Night VFR variants (Mod. No. 125 and Mod. No. 151 respectively), M24C Plus Maintenance Manual 215-00-24C Issue C or later approved revisions.

(c) Maintenance Schedule 045-00-24C Issue G or later approved revision.

For the M24C equipped with the Rotax 915iS (M24C Plus) and Day/Night VFR variants (Mod. No. 125 and Mod. No. 151 respectively), M24C Plus Maintenance Schedule 237-00- 24C Issue B or later approved revision.

- 11.2 The following placards are to be fitted:
 - (a) Engine RPM limits
 - (b) Rotor RPM
 - (c) Loading conditions (placard on instrument panel)
 - (d) Fuel quantity & type (placard next to fuel filler)
 - (e) All switches (engraved on switch or placard next to switch)
 - (f) Occupant warning (placard on instrument panel)
 - (g) Limitations as per Permit to Fly (placard on instrument panel)
 - (h) Engine CT limits (markings on instrument panel)
 - (i) Secondary control functions (placards)
 - (j) Permanent & fireproof attachment of aircraft registration no. & aircraft serial no. (plate affixed on starboard side of the forward keel tube)

See Appendix D for placards fitted as standard.

(12) MANDATORY MODIFICATIONS / MODIFICATIONS / SERVICE BULLETINS / AIRWORTHINESS DIRECTIVES ETC:

Refer to Appendix A for Mandatory Modifications.

For applicable Magni-Gyro Optional Modifications Refer to: Mod.7312 Modification Register.

For applicable Magni-Gyro Service Bulletins Refer to: Mod.7313 Service Bulletin Register.

For applicable AD's and MPD's refer to Mod.7313 Service Bulletin Register.

(13) MINIMUM PERFORMANCE AT MAX TAKE-OFF WEIGHT:

Minimum performance at max take-off weight: 625 fpm at 65 mph

800 fpm at 73 mph (915iS engine) (Addendum 1, Mod.

No 125)

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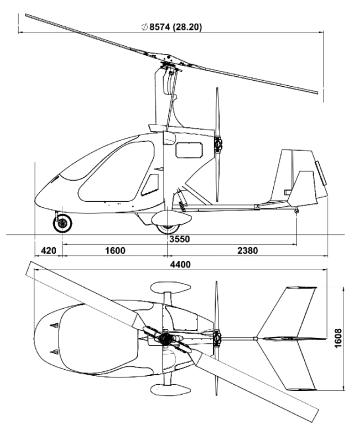
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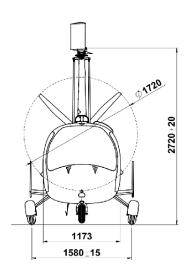
Issue History

Issue No.	Date	Amendment
1	04/10/2010	Initial Issue
2	04/03/2011	Updated Type Design Build Standard report under section 4, reference to MPD 2011-001 under section 12 and updated modifications listing under Appendices A and B.
3	19/05/2011	Addition of limitation for gyroplane training under section 8H. Section 11.1a) Flight Manual status updated.
4	01/08/2011	Maintenance Manual and Maintenance Schedule updated under sections 11.1b) and 11.1c).
5	28/10/2011	Flight Manual, Maintenance Manual and Maintenance Schedule updated under section 11.1.
6	24/06/2015	Manufacturer address. Flight Manual and Maintenance Schedule Issue. Service Bulletins list. Optional Modification list. Pilot Checklist placard.
7	10/02/2016	Manufacturer's address corrected. UK importer and distributor corrected. Propeller type corrected. Optional Modification list updated.
8	09/06/2016	Power Plant max CT instead of max CHT. Flight Manual Issue updated. Optional Modification list updated.
9	23/03/2017	Type Design Build Standard updated to Issue F. Engine Oil Spec updated. Section 11.2) updated. Mandatory Modifications/Service Bulletins/ Airworthiness Directives updated. Mandatory and approved optional modifications updated. Standard placards updated.
10	14/12/2017	Mandatory Modifications/Service Bulletins/ Airworthiness Directives updated. Approved optional modifications updated.
		Service Bulletins list updated. Pilot notes CT updated. Approved optional modifications list updated.
12	03/05/2019	Mandatory Modifications/Service Bulletins/ Airworthiness Directives now listed by cross reference. Instructions for Continued Airworthiness now referencing latest approved revision
13	31/01/2020	Release of 915iS modification (AAN 29276 Addendum 1, Mod. No. 125)
14	25/07/2023	Release of Night VFR modification for M24C Plus (AAN 29276 Addendum 2, Mod. No. 151)

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Illustration of Aircraft Dimensions in millimetres (feet).





Note: the rotor is not in scale

APPENDIX A - MANDATORY MODIFICATIONS

Modification Reference	Description
MOD 039	Rudder pedal mounting block (included in Type Design Build Standard 002- 00-24C issue E)
MOD 101	Coolant temperature markings on Rotax engines

APPENDIX B - APPROVED OPTIONAL MODIFICATIONS

See Mod.7312 Modification Register

APPENDIX C – WEIGHING INFORMATION

N/A Aircraft to be weighed by manufacturer.

APPENDIX D - STANDARD PLACARDS

(Copied from Flight Manual)

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In conformity with BCAR Section T the following placards and markings are installed:

- a. All cockpit controls are clearly marked as to their function and method of operation.
- b. Fuel and oil filler openings are clearly marked, together with the grade or type required. Fuel tank capacity is also clearly marked:

C.

Fuel



M-24 Orion
Max Usable Fuel: 78.5lt
Minimum Grade of Fuel to be Used:
MON 85 RON 95 UNLEADED FUEL
For 100L AVGAS use check Rotax Manual

For 915iS engine (Addendum 1, Mod. No. 125)



M24C Plus

Max Usable Fuel: 78.5 lt

Minimum Grade of Fuel to be Used:

MON 85 RON 95 UNLEADED FUEL

For 100L AVGAS use check Rotax Manual

Oil



For 915iS engine (Addendum 1, Mod. No. 125)



M24C Plus

Oil used in this aircraft:

AeroShell SPORT PLUS 4

Rotax® RON 424

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d. Loading conditions are clearly marked.

ALAGE	VGYRO
M-24	Orion
Max Indicated Airspeed (V _{NE})100mph Min speed for power-on flight30mph Empty Weight297 Kg	MTOM
	ANOEUVERS INVOLVING DELIBERATE REDUCTIOI RCRAFT SHALL BE FLOWN BY DAY VFR ONLY.

For 915iS engine (Addendum 1, Mod. No. 125)

(ALA)	GNGYRO)
M2	4C Plus
Max Indicated Airspeed (V _{NE})100mph Min speed for power-on flight30mph Empty Weight315 Kg	MTOM
AEROBATIC MANOEUVERS ARE PROHIBITED. MAI IN NORMAL "G" SHALL BE AVOIDED. THE AIR	

For the Day/Night VFR variant (Addendum 2, Mod. No. 151)

For the Day/Night VFR Variant (F	Addendum 2, Iviod. INO. 151)
	Plus Night
Max Indicated Airspeed (V _{NE})100mph Min speed for power-on flight30mph Empty Weight317 Kg	MTOM535 Kg Max cockpit loading (including fuel)218 Kg Min solo pilot weight60 Kg CG Limits(refer to flight manual)
AEROBATIC MANOEUVERS ARE PROHIBITED. MAN IN NORMAL "G" SHALL BE AVOIDED. THE AIRCRAF (WITH REQUIRED EQUIPMENT)	OEUVERS INVOLVING DELIBERATE REDUCTION T SHALL BE FLOWN BY DAY AND NIGHT VFR ONLY

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e. Engine limitations are clearly marked:

	Engine	e Parameter	s
	Min	Norm	Max
Temp	erature	(°C)	
OIL	50	90-110	130
CT	50	90-110	120
EGT		750-850	950
Oil Pr	ess (0,	1 x bar)	
	15	20-50	70
Max E	ngine R	PM	5800
Max c	ontinuo	us RPM	5500
Max N	IAP (5 n	nins)39	9.9 inHg
Max N	IAP (co	ntinuous).35	5.4 inHg

For 915iS engine (Addendum 1, Mod. No. 125)

	Engi	ne Parame	eters
Temp	erature	(°C)	
energija Ostavani	Min	Norm	Max
OIL	50	90-110	130
CT	50	90-110	120
EGT		750-850	950
OIL P	RESS	(0,1 x bar)	
		20-50	70
Max E	ngine l	RPM	5800
Max c	ontinue	ous RPM	5500
Max N	IAP	5	1 InHo

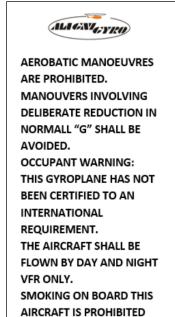
f. Other placards fitted:

ALIGNIGYED) **AEROBATIC MANOEUVERS** ARE PROHIBITED. MANOEUVERS INVOLVING **DELIBERATE REDUCTION** IN NORMAL "G" SHALL BE AVOIDED. OCCUPANT WARNING: THIS GYROPLANE HAS NOT **BEEN CERTIFIED TO AN** INTERNATIONAL REQUIREMENT. THE AIRCRAFT SHALL BE FLOWN BY DAY VFR ONLY. SMOKING ON BOARD THIS AIRCRAFT IS PROHIBITED.

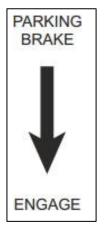
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For the Day/Night VFR variant (Addendum 2, Mod. No. 151)









Baggage Bay – Located beneath co-pilot seat cushion

COMPARTMENT NOT BE USED FOR STOWAGE UNLESS BAGGAGE BAY LINER IS INSTALLED

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Auxiliary Power Sockets (where fitted)

In the event of engine generator failure all equipment powered via the 12v sockets must be turned off.

POWER SOCKET SUPPLY

Output Voltage 12V - Output current 6AMP

Baggage Bay (When liner fitted) – Located beneath co-pilot seat cushion

BAGAGGE BAY LOAD LIMITATIONS

MAXIMUM LOAD CAPACITY 10kg

Strobe/Positioning Lights (If fitted)

WARNING

TURN OFF STROBE LIGHTS WHEN IN THE VICINITY OF OTHER AIRCRAFT OR PERSONS ON THE GROUND.

THE STROBE LIGHTS MUST ALSO BE TURNED OFF PRIOR TO THE LANDING PHASE AND PRIOR TO ACTIVATION THE LANDING LIGHT.

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